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Numerical Optimal Control for Nonsmooth Dynamic Systems

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Continuous-Time Optimal Control Problems (OCP)



Continuous-Time OCP with Ordinary Differential Equation (ODE) Constraints

$$\min_{x(\cdot),u(\cdot)} \int_0^T L_{\mathbf{c}}(x(t),u(t)) \, \mathrm{d}t + E(x(T))$$
s.t.
$$x(0) = \bar{x}_0$$

$$\dot{x}(t) = f(x(t),u(t))$$

$$0 \ge h(x(t),u(t)), \ t \in [0,T]$$

$$0 \ge r(x(T))$$

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Can in most applications assume convexity of all "outer" problem functions: L_c, E, h, r .



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Three levels of difficulty:



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(a) Linear ODE: f(x, u) = Ax + Bu



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- (a) Linear ODE: f(x, u) = Ax + Bu
- (b) Nonlinear smooth ODE: $f \in \mathcal{C}^1$



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Three levels of difficulty:

- (a) Linear ODE: f(x, u) = Ax + Bu
- (b) Nonlinear smooth ODE: $f \in \mathcal{C}^1$
- (c) Nonsmooth Dynamics (NSD):
 - ightharpoonup f not differentiable (NSD1),
 - ightharpoonup f not continuous (NSD2), or even
 - f not finite valued, discontinuous state x(t) (NSD3)



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First focus on smooth cases (a) and (b).



Continuous-Time OCP

$$\min_{x(\cdot),u(\cdot)} \int_0^T L_{\mathbf{c}}(x(t),u(t)) dt + E(x(T))$$
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$$x(0) = \bar{x}_0$$

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Recall: Runge-Kutta Discretization for Smooth Systems



Ordinary Differential Equation (ODE)

$$\dot{x}(t) = \underbrace{f(x(t), u(t))}_{=:v(t)}$$

Initial Value Problem (IVP)

$$x(0) = \bar{x}_0$$

$$v(t) = f(x(t), u(t))$$

$$\dot{x}(t) = v(t)$$

$$t \in [0, T]$$

Discretization: N Runge-Kutta steps of each n_s stages

$$x_{0,0} = \bar{x}_0,$$
 $\Delta t = \frac{T}{N}$
 $v_{k,j} = f(x_{k,j}, u_k)$
 $x_{k,j} = x_{k,0} + \Delta t \sum_{n=1}^{n_s} a_{jn} v_{k,n}$
 $x_{k+1,0} = x_{k,0} + \Delta t \sum_{n=1}^{n_s} b_n v_{k,n}$
 $j = 1, \dots, n_s, \quad k = 0, \dots, N-1$

For fixed controls and initial value: square system with $n_x+N(2n_s+1)n_x$ unknowns, implicitly defined via $n_x+N(2n_s+1)n_x$ equations.

(trivial eliminations in case of explicit RK methods)



Continuous time OCP

$$\begin{aligned} \min_{x(\cdot),u(\cdot)} & \int_0^T L_{\mathbf{c}}(x(t),u(t)) \, \mathrm{d}t + E(x(T)) \\ \text{s.t.} & x(0) = \bar{x}_0 \\ & \dot{x}(t) = f(x(t),u(t)) \\ & 0 \geq h(x(t),u(t)), \ t \in [0,T] \\ & 0 \geq r(x(T)) \end{aligned}$$

Direct methods "first discretize, then optimize"



Continuous time OCP

$$\min_{x(\cdot), u(\cdot)} \int_{0}^{T} L_{c}(x(t), u(t)) dt + E(x(T))$$
s.t. $x(0) = \bar{x}_{0}$

$$\dot{x}(t) = f(x(t), u(t))$$

$$0 \ge h(x(t), u(t)), \ t \in [0, T]$$

$$0 \ge r(x(T))$$

Direct methods "first discretize, then optimize" 1. Parameterize controls, e.g. $u(t) = u_n, t \in [t_n, t_{n+1}].$



Continuous time OCP

$$\min_{x(\cdot),u(\cdot)} \int_0^T L_{\mathbf{c}}(x(t),u(t)) \, \mathrm{d}t + E(x(T))$$
s.t.
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Direct methods "first discretize, then optimize"

- 1. Parameterize controls, e.g. $u(t) = u_n, t \in [t_n, t_{n+1}].$
- 2. Discretize cost and dynamics

$$L_{\mathrm{d}}(x_n, z_k, u_n) \approx \int_{t_n}^{t_{n+1}} L_{\mathrm{c}}(x(t), u(t)) \,\mathrm{d}t$$

Replace
$$\dot{x}=f(x,u)$$
 by
$$x_{n+1}=\phi_f(x_n,z_n,u_n)$$

$$0=\phi_{\mathrm{int}}(x_n,z_n,u_n)$$



Continuous time OCP

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3. Also discretize path constraints

$$0 \ge \phi_h(x_n, z_n, u_n), \ n = 0, \dots N - 1.$$



Continuous time OCP

$$\min_{x(\cdot), u(\cdot)} \int_{0}^{T} L_{c}(x(t), u(t)) dt + E(x(T))$$
s.t. $x(0) = \bar{x}_{0}$

$$\dot{x}(t) = f(x(t), u(t))$$

$$0 \ge h(x(t), u(t)), t \in [0, T]$$

$$0 \ge r(x(T))$$

Direct methods "first discretize, then optimize"

Discrete time OCP (an NLP)

$$\min_{\mathbf{x}, \mathbf{z}, \mathbf{u}} \sum_{k=0}^{N-1} L_{\mathbf{d}}(x_k, z_k, u_k) + E(x_N)$$
s.t. $x_0 = \bar{x}_0$

$$x_{n+1} = \phi_f(x_n, z_n, u_n)$$

$$0 = \phi_{\text{int}}(x_n, z_n, u_n)$$

$$0 \ge \phi_h(x_n, z_n, u_n), \ n = 0, \dots, N-1$$

$$0 \ge r(x_N)$$

Variables $\mathbf{x} = (x_0, \dots, x_N)$, $\mathbf{z} = (z_0, \dots, z_N)$ and $\mathbf{u} = (u_0, \dots, u_{N-1})$. Here, \mathbf{z} are the intermediate variables of the integrator (e.g. Runge-Kutta)

Simplest Direct Transcription: Single Step Explicit Euler

(not recommended in practice, other Runge-Kutta methods are much more efficient)



Continuous time OCP

$$\min_{x(\cdot),u(\cdot)} \int_0^T L_c(x(t),u(t)) dt + E(x(T))$$
s.t.
$$x(0) = \bar{x}_0$$

$$\dot{x}(t) = f(x(t),u(t))$$

$$0 \ge h(x(t),u(t)), t \in [0,T]$$

$$0 \ge r(x(T))$$

Direct methods: first discretize, then optimize

Single Step Explicit Euler NLP, with $\Delta t = \frac{T}{N}$

$$\min_{\mathbf{x}, \mathbf{u}} \sum_{k=0}^{N-1} L_{c}(x_{k}, u_{k}) \Delta t + E(x_{N})$$
s.t. $x_{0} = \bar{x}_{0}$

$$x_{n+1} = x_{n} + f(x_{n}, u_{n}) \Delta t$$

$$0 \ge h(x_{n}, u_{n}), \ n = 0, \dots, N-1$$

$$0 \ge r(x_{N})$$

Variables $\mathbf{x} = (x_0, \dots, x_N)$ and $\mathbf{u} = (u_0, \dots, u_{N-1})$. (single step explicit Euler has no internal integrator variables \mathbf{z})

Sparse NLP resulting from direct transcription



Discrete time OCP (an NLP)

$$\min_{\mathbf{x}, \mathbf{z}, \mathbf{u}} \sum_{k=0}^{N-1} L_{d}(x_{k}, z_{n}, u_{k}) + E(x_{N})$$
s.t. $x_{0} = \bar{x}_{0}$

$$x_{n+1} = \phi_{f}(x_{n}, z_{n}, u_{n})$$

$$0 = \phi_{int}(x_{n}, z_{n}, u_{n})$$

$$0 \ge \phi_{h}(x_{n}, z_{n}, u_{n}), \ n = 0, \dots, N-1$$

$$0 \ge r(x_{N})$$

Variables $w = (\mathbf{x}, \mathbf{z}, \mathbf{u})$

Nonlinear Program (NLP)

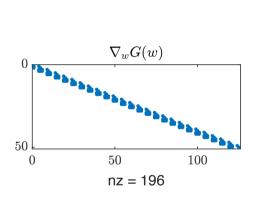
$$\min_{w \in \mathbb{R}^{n_x}} F(w)$$
s.t. $G(w) = 0$

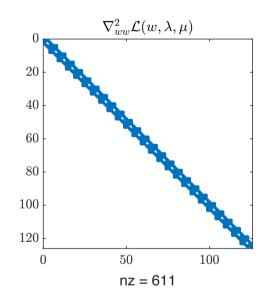
$$H(w) \ge 0$$

Large and sparse NLP

Sparse NLP resulting from direct transcription







Variables $w = (\mathbf{x}, \mathbf{z}, \mathbf{u})$

Nonlinear Program (NLP)

$$\min_{w \in \mathbb{R}^{n_x}} F(w)$$
s.t. $G(w) = 0$

$$H(w) \ge 0$$

Large and sparse NLP

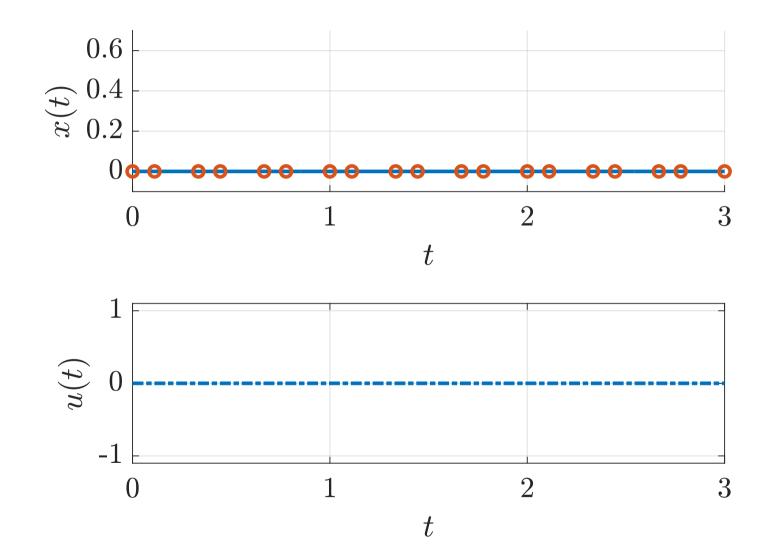
Illustrative example of direct collocation with Newton-type optimization

Illustrative nonlinear optimal control problem (with one state and one control)

- choose N=9 equal intervals and Radau-IIA collocation with $n_s=2$ stages
- \blacktriangleright obtain nonlinear program with $n_x + (2n_s + 1)Nn_x + Nn_u$ variables
- initialize with zeros everywhere, solve with CasADi and Ipopt (interior point)

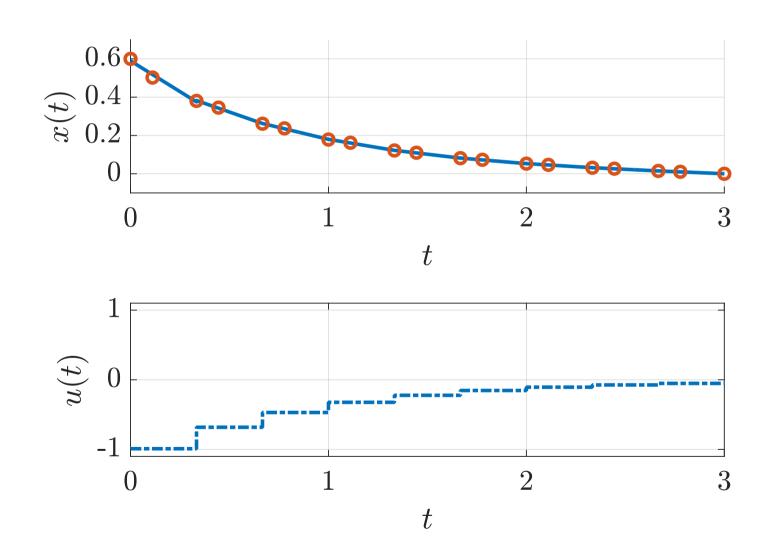
Illustrative example: Initialization





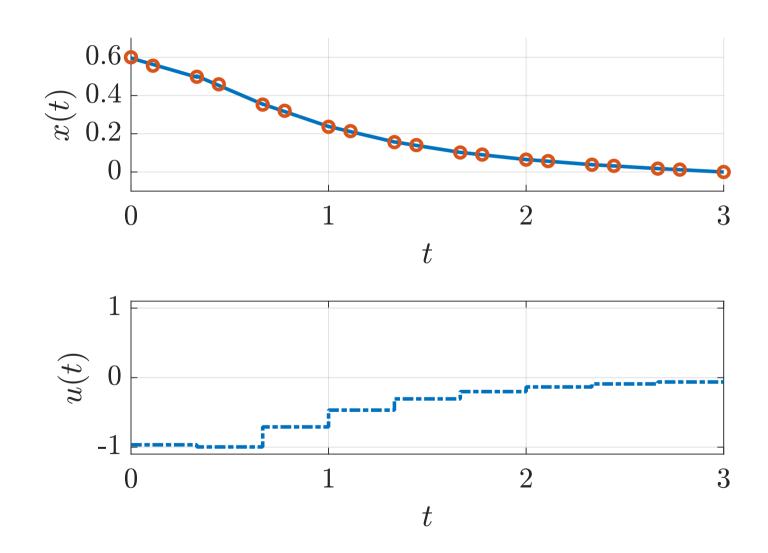
Illustrative example: First Iterate





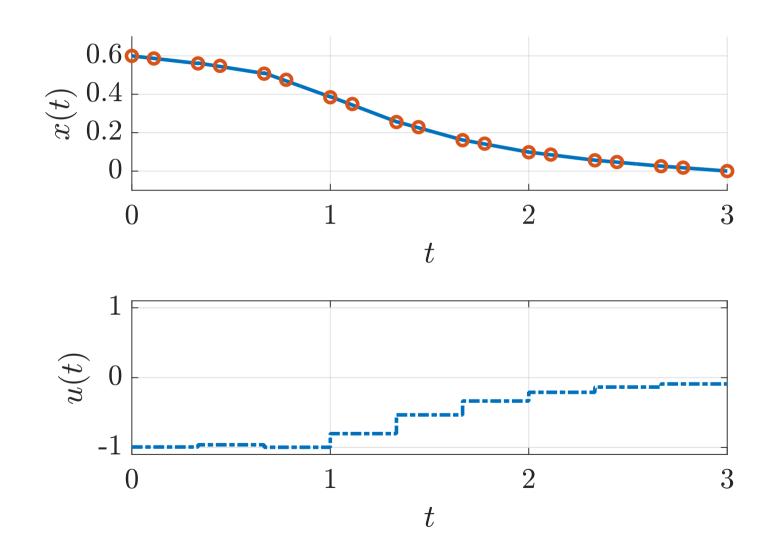
Illustrative example: Second Iterate





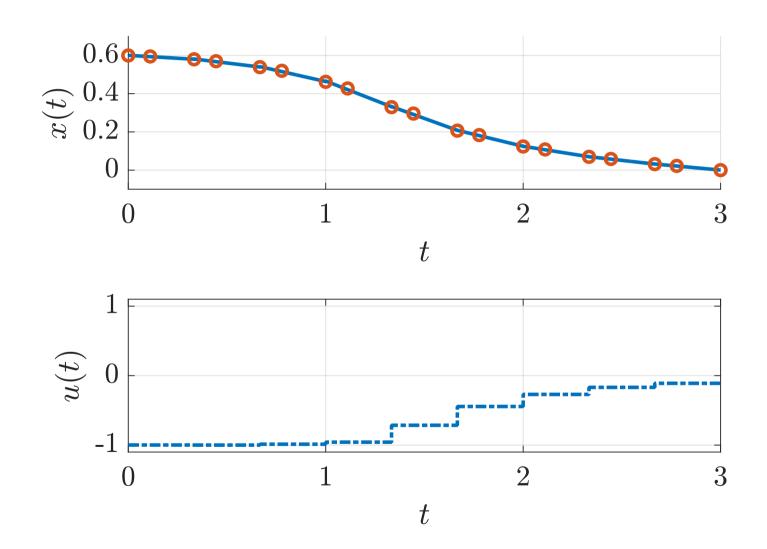
Illustrative example: Third Iterate





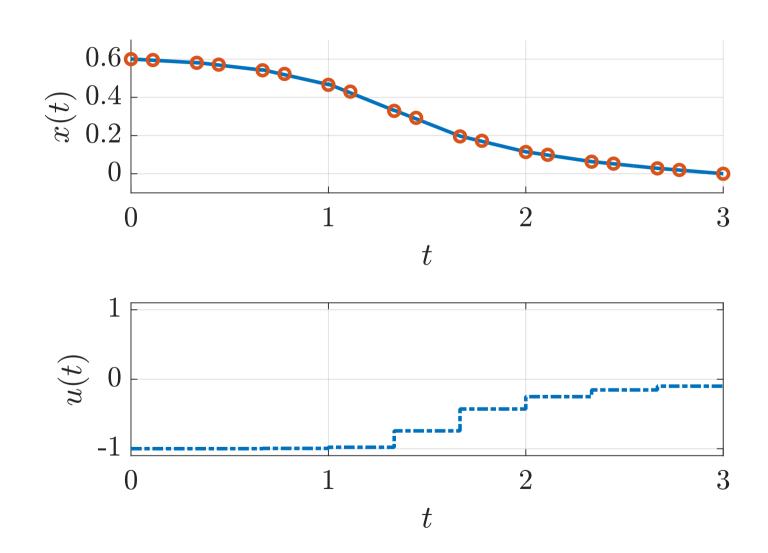
Illustrative example: Fourth Iterate





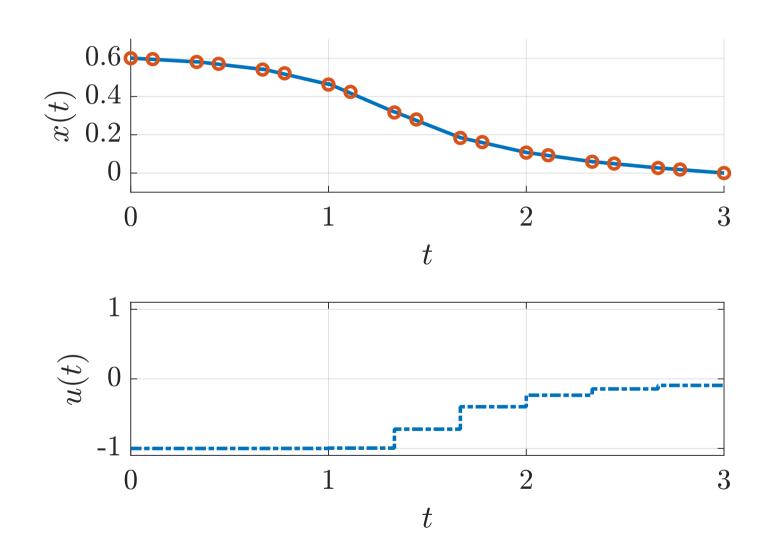
Illustrative example: Fifth Iterate





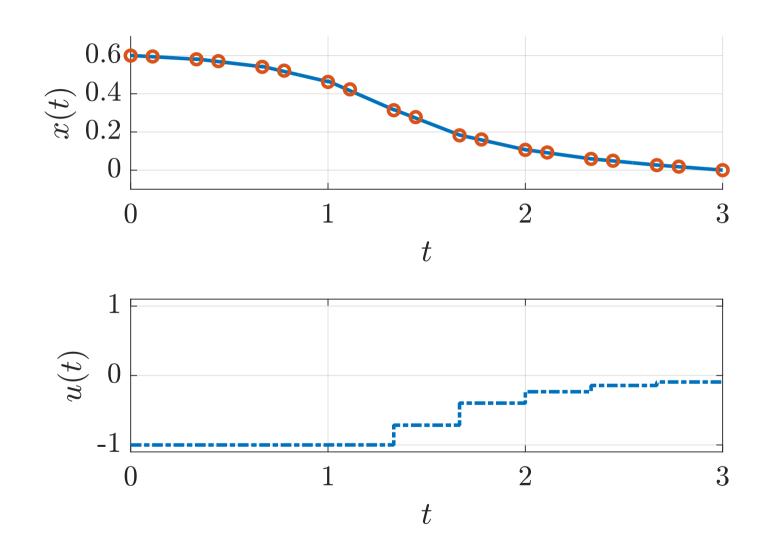
Illustrative example: Sixth Iterate





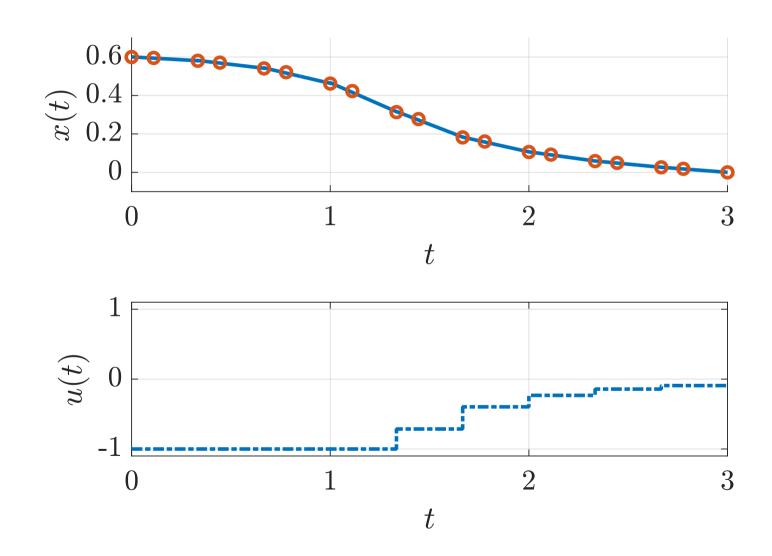
Illustrative example: Seventh Iterate





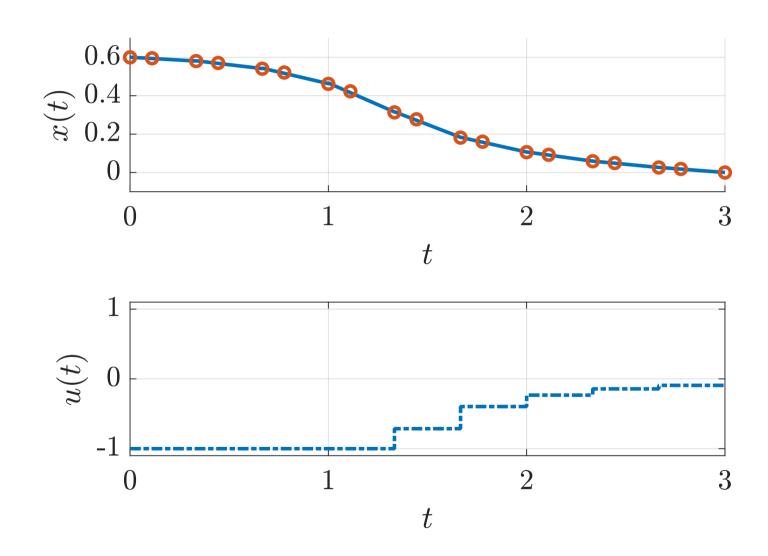
Illustrative example: Eighth Iterate





Illustrative example: Solution after Nine Newton-type Iterations





More Complex Example: Power Optimal Trajectories in Airborne Wind Energy (AWE) formulated and solved daily by practitioners using open-source python package "AWEBox" [De Schutter et al. 2023]





For simple plane attached to a tether:

- · 20 differential states (3+3 trans, 9+3 rotation, 1+1 tether)
- 1 algebraic state (tether force)
- · 8 invariants (6 rotation, 2 due to tether constraint)
- · 3 control inputs (aileron, elevator, tether length)

Translational:
$$\begin{bmatrix} m & 0 & 0 & x \\ 0 & m & 0 & y \\ 0 & 0 & m & z \\ x & y & z & 0 \end{bmatrix} \begin{bmatrix} \ddot{x} \\ \ddot{y} \\ \ddot{z} \\ \lambda \end{bmatrix} = \begin{bmatrix} F_x + m \left(\dot{\delta}^2 r_A + \dot{\delta}^2 x + 2 \dot{\delta} \dot{y} + \ddot{\delta} y \right) \\ F_y + m \left(y \dot{\delta}^2 - 2 \dot{x} \dot{\delta} - \ddot{\delta} (rA + x) \right) \\ F_z - gm \\ -\dot{x}^2 - \dot{y}^2 - \dot{z}^2 \end{bmatrix}$$

Rotational:
$$\dot{R} = R\omega_{\times} - R^{T} \begin{bmatrix} 0 \\ 0 \\ \dot{\delta} \end{bmatrix}, \quad J\dot{\omega} = T - \omega \times J\omega, \quad R = \begin{bmatrix} \vec{E}_{x} & \vec{E}_{y} & \vec{E}_{z} \end{bmatrix}$$

Aero. coefficients:
$$\vec{v} = \begin{bmatrix} \dot{x} - \dot{\delta}y \\ \dot{y} + \dot{\delta}(r_{\rm A} + x) \\ \dot{z} \end{bmatrix} - \vec{w}(x, y, z, \delta, t), \qquad \alpha = -\frac{\vec{E}_z^T \vec{v}}{\vec{E}_x^T \vec{v}}, \qquad \beta = \frac{\vec{E}_y^T \vec{v}}{\vec{E}_x^T \vec{v}}$$

Aero. forces/torques:
$$\vec{F}_{A} = \frac{1}{2}\rho A \|\vec{v}\| (C_{L}\vec{v} \times \vec{E}_{y} - C_{D}\vec{v}), \quad \vec{T}_{A} = \frac{1}{2}\rho A \|\vec{v}\|^{2} \begin{bmatrix} C_{R} \\ C_{P} \\ C_{Y} \end{bmatrix}$$

Newton-Type Optimization Iterations for Power Optimal Flight (video by Greg Horn, using CasADi and Ipopt as optimization engine)



