

## Durable structure reinforcements for ram-air kites

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## Who am I?



#### Marie Curie PhD student within **AWESCO** network

- Background in Aerospace Engineering from TU Delft and DTU
- Started PhD at SkySails GmbH in Hamburg until their insolvency in April 2016
- Uni Freiburg is new employer since July 2016
- Using office of SkySails Group in Hamburg to continue my research





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How to support the structure such that it will live longer?







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#### REINFORCEMENTS



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#### **Ram-air kites**

#### **Structural layout**

- Canopy the kite's skin
- Ribs internal structure which defines aerodynamic shape and stiffness
- Bridles connecting lower canopy with tether, also used for steering
- KCU (Kite control unit) steering, wind speed measurement, IMU, etc.
- Inlet Hole at leading edge for inflation
- Tether attached to ground station or vehicle/pilot







- + Cheap, lightweight, easy to storage
- Durability issues due to UV light exposure, fatigue, and storage
- Lower CL/CD compared to rigid wings
- Difficult to model due to large deformations





#### What are reinforcements?



## Structural support to strengthen highly loaded areas



### What are reinforcements?



# Structural support to strengthen highly loaded areas

- Example: 400m2 kite produces approximately 400kN tether force
  - The load acting on the centre rib is roughly 40kN (2 SUVs)
  - Fabric thickness is 1 mm
- Use fabric tape to direct loads from the top canopy to the bridle points



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#### How to model the structure



Strong equilibrium form of an elastostatic body

$$\nabla \cdot \boldsymbol{\sigma} - \mathbf{b} = \mathbf{0} \quad \text{in} \quad \Omega$$
$$\mathbf{u} = \mathbf{0} \quad \text{on} \quad \Gamma_0$$
$$\boldsymbol{\sigma}^{\mathbf{T}} \mathbf{n} = \mathbf{t} \quad \text{on} \quad \Gamma_t$$

**Kinematics** 

$$\boldsymbol{\epsilon} = \nabla \cdot \mathbf{u}$$

Constitutive relations

$$\boldsymbol{\sigma} = \mathbf{C} \; \boldsymbol{\epsilon}$$



#### How to model the structure



Weak equilibrium form of an elastostatic body

$$\Pi = \int_{\Omega} \nabla \cdot \sigma^{\mathbf{T}} \mathbf{u} \ dV - \int_{\Omega} \mathbf{b}^{\mathbf{T}} \mathbf{u} \ dV$$

Variational principle of total potential energy

 $\delta \Pi = \mathbf{0}$ 

Applying Green's Theorem

$$\int_{\Omega} \sigma^{\mathbf{T}} \delta \epsilon \, dV = \int_{\Omega} \mathbf{b}^{\mathbf{T}} \delta \mathbf{u} \, dV + \int_{\Gamma_t} \mathbf{t}^{\mathbf{T}} \delta \mathbf{u} \, dS$$



How to model the structure

Finite element approximation with shape functions

 $\mathbf{u}_e = \mathbf{N} \; \mathbf{d}_e$ 

Static equilibrium equation

 $\mathbf{K}\mathbf{d}=\mathbf{F_{ext}}$ 

Dynamic equation of motion

 $\mathbf{M}\ddot{\mathbf{d}} + \mathbf{C}\dot{\mathbf{d}} + \mathbf{K}\mathbf{d} = \mathbf{F_{ext}}$ 





## How to model the reinforcements



#### Patch

- Introduce continuum elements with different material properties and/or thickness
- + Reinforcement has bending and shear resistance
- more elements introduced
- re-meshing required when reinforcements are moved





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## How to model the reinforcements



 Introduce truss elements with different material properties and/or thickness (Taylor et al., 2005)

$$\begin{aligned} \mathbf{x}^{i} &= \xi_{1}^{i} \; \tilde{\mathbf{x}}^{1} + \xi_{2}^{i} \; \tilde{\mathbf{x}}^{2} + \xi_{3}^{i} \; \tilde{\mathbf{x}}^{3} \\ \xi_{2}^{i} &= \frac{||\mathbf{X}^{i} - \mathbf{X}^{1}||}{||\mathbf{X}^{2} - \mathbf{X}^{1}||} \quad ; \quad \xi_{1}^{i} = 1 - \xi_{2}^{i} \quad ; \end{aligned}$$

- + no re-meshing
- + no additional degrees of freedom
- reinforcements have no bending and shear resistance
- might cause instabilities if reinforcement stiffness is too high

 $\xi_3^i$ 

X,







### How to model the reinforcements



**Reinforcement applied to the rib structure** 





#### **Computational Fluid Dynamics** (CFD)

- Solve the Navier-Stokes equations with, e.g. finite volume approach
   → Obtain pressure and fluxes in each finite volume
- Kites fly at larger Reynold's number (Re > 1e6) and eddies cannot be resolved
  → turbulence model required
- Kites fly at large angle of attack to maximise tether force

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 $\rightarrow$  flow separation can be expected







## How to model the flow



#### **Computational Fluid Dynamics** (CFD)

- From the fluid mesh integrate pressure to nodal forces
- Alternative to CFD is potential flow (panel method) which is much faster but has its limitations





## How to couple both models





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## **Hron-Turek benchmark**



#### Waving flag (Hron & Turek, 2006)

- Virtual wind tunnel with parabolic inflow conditions
  - > Re = 200
  - > Density ratio = 1
  - ≻ E = 1.6 MPa
- Mesh motion and interface interpolation with radial basis functions
- Parabolic inflow conditions and slight asymmetry to induce vortex shedding



#### **Hron-Turek benchmark**



#### Waving flag (Hron & Turek, 2006)





#### For the first publication:

- 2D reinforcement layout optimisation
- How to get from 3D to 2D?
  - Assume constant pressure over single kite cell
  - Add internal pressure
- Static loads only
- Trim kite in each iteration by finding
  X\_trim such that moment is zero
- What to optimise?

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#### **Optimisation statement**

maximise  $\theta_{,\alpha}$   $f(\theta, \alpha) = C_r(1 + E^2)$ subject to  $C_m = 0$   $\theta_{min} \le \theta \le \theta_{max}$   $\alpha_{min} \le \alpha \le \alpha_{max}$ 

with 
$$Cr = \sqrt{(C_l)^2 + (C_d)^2}$$
;  $E = C_l/C_d$ 





#### **Optimisation statement 2**

 $\begin{array}{ll} \underset{\theta,\alpha}{\text{maximise}} & f(\theta,\alpha) = \beta \ C_r(1+E^2) + (1-\beta) \ 1/(\mathbf{u^T K u}) \\ \text{subject to} & C_m = 0 \\ & \theta_{min} \leq \theta \leq \theta_{max} \\ & \alpha_{min} \leq \alpha \leq \alpha_{max} \\ & \sigma_{\text{VM}} \leq \sigma_{\text{VM}_{\text{max}}} \end{array}$ 

with 
$$Cr = \sqrt{(C_l)^2 + (C_d)^2}$$
;  $E = C_l/C_d$ 





#### Questions

- What solver can be used?
- Gradient or non-gradient based?
- How to parameterise the design variables?







#### Thank you for your attention!



### CFD vs. potential flow







### CFD vs. potential flow



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#### **Deformed mh92 airfoil**





## No reinforcements









#### With reinforcements



